

For the following multiple choice questions, choose the *most correct* answer. For show the detailed solution for each question to guarantee the grade. (2 points)

[1-2]: In center of pressure experiment, if the plane surface is partially immersed, the water level the width of immersed surface $b=7.5$ cm. ($\gamma_{water} = 9810 \text{ N/m}^3$). Answer Problems (1-2):

- 1) The hydrostatic pressure force on the plane surface is:

- a. 0.59 N
- b. 0.33 N
- c. 0.92 N
- d. 0.15 N
- e. None of the above

- 2) The theoretical center of pressure measured from the surface of the water is:

- a. 3.33 cm
- b. 1.33 cm
- c. 2.67 cm
- d. 2.00 cm
- e. None of the above

- 3) Thermal conductivity of a material is

- a. The resistance of a material to conduct heat through
- b. The ability of a solid material to store heat
- c. The ability of a material to conduct heat
- d. A measure of liquids ability to convect heat
- e. All of the above

[4-5] In the losses experiment, if the following data were measured: mass flow rate of 0.2 kg/s, density of water is $\rho = 1000 \text{ kg/m}^3$, diameter of small pipe size 14 mm, dynamic viscosity of water is $\mu = 1 \times 10^{-3} \text{ N.s/m}^2$, roughness of the pipe surface is $\epsilon = 0.0015 \text{ mm}$. Answer Problems (4-5):

- 4) The Reynolds number is:

- a. 2100.12
- b. 18189.14
- c. 57142.86
- d. 36378.27
- e. None of the above

- 5) The friction factor is:

- a. 0.018
- b. 0.034
- c. 0.043
- d. 0.027
- e. None of the above

6) In impact of water jet experiment, the water density is 1000 kg/m^3 , the mass flow rate is 0.4 kg/s , the height of vane above nozzle outlet is 0.04 m and the diameter of nozzle is 0.01 m . If a hemispherical cap is used, the theoretical water jet force is:

- a. 6.30 N
- b. 4.01 N
- c. 7.64 N
- d. 5.10 N
- e. None of the above

7) In "Flow through a nozzle" experiment, one of the following statements is correct:

- a. Throat pressure is minimum pressure reading inside the nozzle
- b. Throat pressure is maximum pressure reading inside the nozzle
- c. Mass flow rate is minimum if the nozzle is choked
- d. Throat pressure is the gage pressure reading of the air supply tank
- e. None of the above

8) Only one of the following statement is correct with regards to the Flow through a nozzle experiment:

- a. As pressure increases in the direction of the flow in the nozzle, velocity decreases
- b. Both pressure and velocity decrease through the nozzle
- c. Mass flow rate of the air increases as the area of the nozzle decreases
- d. Cross section area of the nozzle increases in the direction of the flow
- e. As the velocity increases in the direction of the flow, pressure decreases

9) In flow through a nozzle experiment, the stagnation "chest" absolute pressure is 290 kPa , the stagnation "chest" temperature is 18°C , the air gas constant is 0.287 kJ/kg.K , the air specific heat ratio is 1.4 , the nozzle throat area is $9.16 \times 10^{-6} \text{ m}^2$ and the throat absolute pressure is 265 kPa . The mass flow rate at the nozzle throat is:

- a. $3.63 \times 10^{-3} \text{ kg/s}$
- b. $2.92 \times 10^{-3} \text{ kg/s}$
- c. $1.84 \times 10^{-3} \text{ kg/s}$
- d. $2.34 \times 10^{-3} \text{ kg/s}$
- e. None of the above

10) In losses in pipes experiment, pressure change in globe valve is measured using:

- | | | | | |
|--------------------------------|--------------------------------|---------------------|----------------------|----------------------|
| a. Pressurized piezometer tube | b. Piezoelectric gage pressure | c. U-tube manometer | d. Pitot-static tube | e. None of the above |
|--------------------------------|--------------------------------|---------------------|----------------------|----------------------|

11) The type of the nozzle used in the "flow through a nozzle" experiment is:

- | | | | | |
|-----------------------|------------------------|-------------------------|------------------------|----------------------|
| a. Divergent-Parallel | b. Convergent-Parallel | c. Convergent-divergent | d. Divergent-divergent | e. None of the above |
|-----------------------|------------------------|-------------------------|------------------------|----------------------|

12) In thermal conductivity experiment, the copper specimen diameter is 0.25 m, the copper specimen length is 50 mm, the water density is 1000 kg/m^3 , the specific heat of water is $4.18 \text{ J/kg}^\circ\text{C}$, the mass flow rate is $2.75 \times 10^{-3} \text{ kg/s}$, the inlet and outlet water temperatures are 15°C and 25°C respectively. The copper specimen ends temperatures are 362.5°C and 170°C . The rate of heat transfer through the copper specimen is:

- a. 34.49 W
- b. 22.99 W
- c. 28.74 W
- d. 17.24 W
- e. None of the above

13) In pump characteristic experiment, for the centrifugal pump the following data were recorded: the suction pressure is 0 bar, the delivery pressure is 0.7 bar, the volume flow rate is $3.20 \times 10^{-3} \text{ m}^3/\text{s}$, the water density is 1000 kg/m^3 , the spring load is 17.64 N, the motor speed is 15 rev/s and the torque arm radius is 0.15 m. The overall efficiency of the pump is:

- a. 0.742
- b. 0.648
- c. 0.901
- d. 0.514
- e. None of the above

14) In "comparison of pump characteristic" experiment one of the following statements is correct:

- a. For a positive displacement reciprocating pump, the amount of fluid flow rate is independent of pump rotational speed.
- b. Pumps extract energy from the fluid passing through.
- c. Pressure of the fluid at the exit of the pump is lower than the pressure of the fluid at the inlet of the pump.
- d. The performance of the pump is measured using coefficient of performance.
- e. None of the above.

15) In "Liquid-vapor saturation curve" experiment only one statement of the following is correct:

- a. Saturation pressure and temperature are independent from each other.
- b. Saturation pressure is the pressure at which the liquid changes phase into super-heated phase.
- c. Saturation temperature is the temperature at which the liquid becomes compressed liquid.
- d. Saturation temperature varies as pressure varies.
- e. None of the above.

Equations sheet:

$$Re = \frac{\rho V D}{\mu}, \bar{m} = \rho V A, K = {}^{\circ}\text{C} + 273, P_{abs} = P_{atmos} + P_{atm}, (dT/dP)_{sat} = v_f/g T/h_{fg}, g = 9.81 \text{ m/s}^2$$

$$h_L = h_f + \sum h_m, h_m = K \frac{V^2}{2g}, h_f = f \frac{1}{D} \frac{V^2}{2g} \quad \text{Laminar: } f = \frac{64}{Re}, \quad \text{Turbulent: } \frac{1}{f} = -1.8 \log \left[\frac{c_0}{Re} + \left(\frac{c_1}{\sqrt{f}} \right)^{1.11} \right]$$

$$\bar{p} = \rho g h_c, \quad F = \rho g h_c A, \quad y_{cp} = y_c + \frac{h_{sat}}{y_c A}, \quad \text{For water: } \rho = 1000 \text{ kg/m}^3$$

$$F = \bar{m}(u_o - u_i \cos \beta), u_o^2 = u^2 - 2gs \quad \bar{m} = \rho u A \quad \text{Bernoulli equation: } \frac{p}{\rho g} + \frac{v^2}{2g} + z = \text{constant}$$

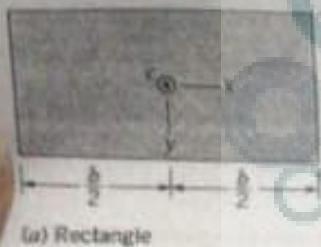
$$\dot{Q} = \bar{m} c (T_{out} - T_{in}), \dot{Q} = -k A \frac{dT}{dx}$$

$$\frac{P}{P_e} = \left(\frac{2}{\gamma+1} \right)^{\frac{\gamma}{\gamma-1}}, v_t = \sqrt{\frac{2 \gamma R T_0}{(\gamma-1)} \left[1 - \left(\frac{P}{P_e} \right)^{\frac{\gamma-1}{\gamma}} \right]}, \quad m_t = A_t P_e \left(\frac{P_1}{P_e} \right)^{1/\gamma} \sqrt{\frac{2 \gamma}{(\gamma-1) R T_0} \left[1 - \left(\frac{P_1}{P_e} \right)^{\frac{\gamma-1}{\gamma}} \right]}, \quad \gamma = \frac{c_p}{c_v}$$

$$h_p = \frac{\Delta p}{\rho g} \times 10^5, \Delta p = p_d - p_s, P_{water} = \rho g Q h_p \times 10^{-3}, P_{brake} = 2 \pi \omega F R \times 10^{-3}, \eta_o = \frac{P_{water}}{P_{brake}}, \eta_g = \frac{Q}{Q_e}$$

$$\dot{Q}_c = \frac{0.75}{12.5} \times 10^{-3} \omega_{pump}, \omega_{pump} = 2 \omega_m \text{ and } \bar{m} = \rho \dot{Q} \quad m_a = 0.00105 \sqrt{\frac{P_1 P_2}{T_1}}, \dot{Q}_H - \dot{W}_c - \dot{W}_F = \dot{Q}_L$$

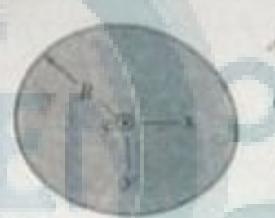
$$COP_{HP} = \frac{Q_H}{Q_H - Q_L}, \quad COP_{HP,rev} = \frac{1}{1 - T_L/T_H}, \quad COP_R = \frac{Q_L}{Q_H - Q_L}, \quad COP_{R,rev} = \frac{1}{T_H/T_L - 1}$$



(a) Rectangle

$$A = ba \\ I_{xc} = \frac{1}{12} ba^3 \\ I_{yc} = \frac{1}{12} ab^3 \\ I_{yz} = 0$$

(b) Circle



$$A = \pi R^2 \\ I_{xc} = I_{yc} = \frac{\pi R^4}{4} \\ I_{yz} = 0$$

Saturated Water and Steam Tables

Pressure bar	Temperature °C	(v _{fg}) m ³ /kg	(h _{fg}) kJ/kg
1	99.60	1.6940	2258
2	120.2	0.8856	2202
3	133.5	0.6057	2164
4	143.6	0.4623	2134
5	151.8	0.3748	2109
6	158.8	0.3156	2087
7	165.0	0.2728	2067
8	170.4	0.2403	2048

⑥ Impact of Water jet

$$\rho = 1000 \text{ kg/m}^3; \dot{m} = 0.4 \frac{\text{kg}}{\text{s}}; S = 0.04 \text{ m}; D_{\text{Nozzle}} = 0.01 \text{ m}$$

→ hemispherical cup

$$F_{\text{th}} = 2 \dot{m} u_0$$

$$= 2(0.4) (5.0153)$$

$$= 4.01224 \text{ N}$$

$$\approx 4.01 \quad \textcircled{b}$$

$$\dot{m} = \rho u A_{\text{nozzle}}$$

$$0.4 = (1000) u * \frac{\pi}{4} (0.01)^2$$

$$u = 5.093 \text{ m/s}$$

$$u_0^2 = u^2 - 2gS \Rightarrow u_0 = \sqrt{5.093^2 - 2(9.81)(0.04)}$$

$$* u_0 = 5.0153 \text{ m/s}$$

⑦ Flow through Nozzle

⑧ c.

a.

⑨ $P_0 = 290 \text{ kPa}$; $T_0 = 18^\circ\text{C}$; $R = 0.287 \frac{\text{kJ}}{\text{kg K}}$; $\gamma = 1.4$

$A_t = 9.16 \times 10^{-4} \text{ m}^2$, $P_t = 265 \text{ kPa}$

$$\dot{m} = A_t P_0 \left(\frac{P_t}{P_0} \right)^{\frac{1}{\gamma-1}} \left(\frac{283}{(18-1)R T_0} \right) \left(1 - \left(\frac{P_t}{P_0} \right)^{\frac{\gamma-1}{\gamma}} \right)$$

$$= (9.16 \times 10^{-4}) \uparrow \left(\frac{265}{290} \right)^{\frac{1}{1.4}} * \sqrt{\frac{2(1.4)}{(1.4-1) 287 \times (18+273)}} * \left(1 - \left(\frac{265}{290} \right)^{\frac{1.4-1}{1.4}} \right)$$

$$= 3.63 \times 10^{-3} \text{ kg/s}$$

⑪ b. convergent - parallel

①

Q & Q : losses on pipes

$$\rho = 0.2 \text{ kg/s}, \rho_0 = 1 \text{ kg/m}^3, D_{\text{wall}} = 14 \text{ mm}, \mu = 1 \times 10^{-3} \frac{\text{Ns}}{\text{m}^2}$$

$$C = 0.0015 \text{ m}$$

$$Q = Re \cdot L \cdot \frac{1}{8} \left(\frac{V}{D} \right) D_{\text{wall}} = 100 \text{ l/s}$$

$$Re = 18184.14$$

Turbulent flow

$$(5) \frac{1}{\sqrt{f}} = -2.0 \log \left(\frac{2.2 \times 10^{-2}}{Re} + \frac{2.51}{Re^{0.5}} \right)$$

$$f = 0.027$$

$$d = 0.02 \text{ m}$$

$$W: \rho = 0.25 \text{ kg/s}, \rho_0 = 1 \text{ kg/m}^3, D_{\text{wall}} = 14 \text{ mm}, \mu = 1 \times 10^{-3} \frac{\text{Ns}}{\text{m}^2}, h_m = 0.1 \text{ m}$$

$$h_m = K \frac{V^2}{2g}$$

$$0.1 = K \frac{(0.624)^2}{2 \times 10}$$

$$h_m = \rho V^2 K$$

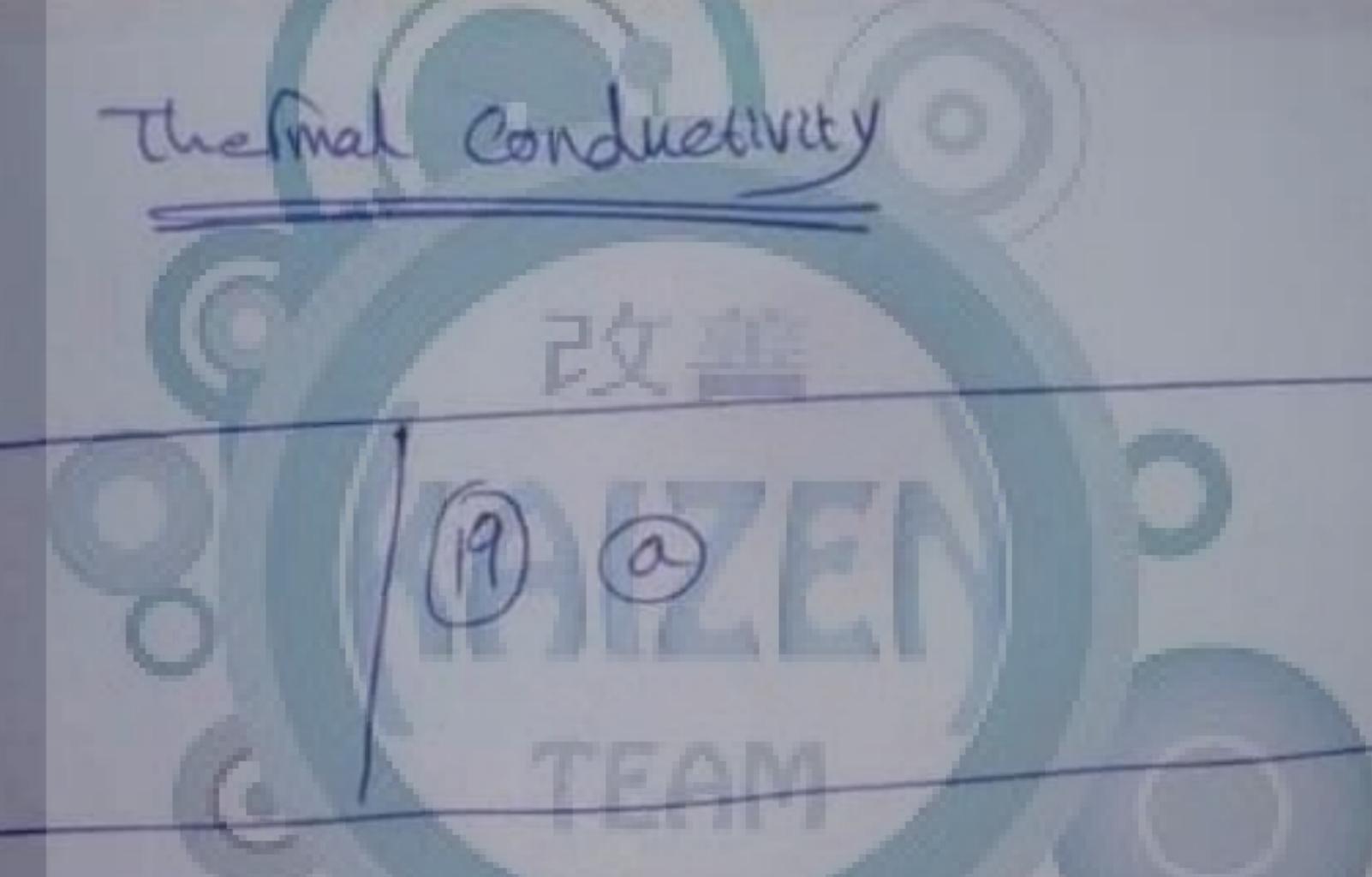
$$0.25 = (100) V + \frac{1}{4} (100)^2$$

$$* V = 1.624 \text{ m/s}$$

$$K = 0.744$$

③ c

⑫ X



Fluid characteristics

→ constant fluid

(13)

$$P_s = 0 \text{ bar} \quad , \quad P_d = 0.7 \text{ bar} , \quad Q = 3.2 \times 10^{-3} \frac{\text{m}^3}{\text{s}} , \quad \rho = 1000 \frac{\text{kg}}{\text{m}^3}$$

$$\Rightarrow F_s = 17.64 \text{ N} , \quad w = 15 \text{ rev/s} , \quad R = 0.15 \text{ m}$$

$$\begin{aligned} D_o &= \frac{P_w}{P_d} \\ &= \frac{0.224}{0.249} \\ &= 0.899 \\ &\approx 0.901 \end{aligned}$$

$$3) P_{water} (\text{kW}) = \rho g Q h_p * 10^{-3}$$

$$\begin{aligned} h_p &= \frac{\Delta P}{\rho g} * 10^3 = \frac{(P_d - P_s) * 10^5}{\rho g} \\ &= \frac{0.7 \text{ bar} * 10^5}{(1000)(9.81)} = \underline{\underline{7.135 \text{ m}}} \end{aligned}$$

$$\begin{aligned} P(\text{kW}) &= (0.001) (9.81) (3.2 \times 10^{-3}) (7.135) * 10^{-3} \\ &\approx 0.224 \end{aligned}$$

$$P_B (\text{kW}) = 2 \pi w n F R * 10^{-3}$$

$$= 2\pi (15) (17.64) (0.15) * 10^{-3}$$

$$= \underline{\underline{0.249}}$$

$$(17) P_s = 0 \text{ bar} , \quad P_d = 0.4 \text{ bar} , \quad Q = 1 \times 10^{-3} \frac{\text{m}^3}{\text{s}} , \quad \rho = 1000 \frac{\text{kg}}{\text{m}^3} , \quad F = 17.64 \text{ N}$$

$$w = 17.64 \text{ N} \quad , \quad R = 0.15 \text{ m}$$

$$\begin{aligned} D_o &= \frac{Q_n}{Q_c} \\ &= \frac{1 \times 10^{-3}}{2.04 \times 10^{-3}} \\ &= \underline{\underline{0.49}} \quad \text{d} \end{aligned}$$

$$\left. \begin{aligned} Q_c &= \left(\frac{0.75}{12.5} \right) * 10^{-3} * w R h_p , \quad w_p = 2 \text{ rev/m} \\ &= 2(17) \\ &= \left(\frac{0.75}{12.5} \right) * 10^{-3} * 34 \\ &= 2.04 \times 10^{-3} \frac{\text{m}^3}{\text{s}} \end{aligned} \right\}$$

(5)

⑧ Heat Pump

$$\dot{Q}_{H} = 1.9 \text{ kW}, \dot{Q}_{L} = 1.4 \text{ kW}, T_H = 60^\circ\text{C}, T_L = 7^\circ\text{C}$$

$$CoP_{(HP)} = \frac{\dot{Q}_F}{\dot{Q}_H - \dot{Q}} = \frac{1.9}{1.9 - 1.4} = 3.8 \quad (b)$$

⑯ Liquid-Vapor

(d)

$$\begin{aligned} \textcircled{2a} \quad \left(\frac{dT}{dp}\right)_{sat} &= T \frac{v_f g}{h_f g} \quad [K/kPa] \\ &= (1335 + 273) K \cdot \frac{T \cdot (0.6057) m^3}{kg \cdot 2164 \frac{kJ}{kg}} \end{aligned}$$

$$*\left(\frac{dT}{dp}\right)_{sat} = 0.114 \frac{K}{kPa}$$